Fall 2021 ME424 Modern Control and Estimation

Lecture Note 4 Stability, Controllability, and Observability

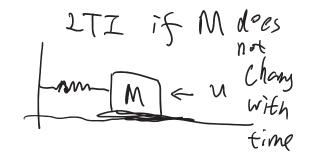
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- So far, we have learned:
 - What is a state space model
 - How to derive state space model from physical systems, nonlinear model, continuous time model, and transfer function model
 - How to identify state space model from input-out data pairs
- Question: how to use a state-space model?
 - Predict system output: solution to a state space model
 - Analysis of behavior: stability/controllability/observability
 - Design controller
- The goal of this lecture note: Given a state space model
 - Derive its <u>solution analytically</u>
 - Stability
 - Controllability <

- State Space Solutions
- Internal Stability
- Controllability
- Observability
- Invariance under Similarity Transformation

General state space model:



 $\begin{aligned} x(k+1) &= \underline{A(k)}x(k) + \underline{B(k)}u(k), \\ y(k) &= \underbrace{C(k)}x(k) + \underbrace{D(k)}u(k) \end{aligned}$

- If system matrices (A(k), B(k), C(k), D(k)) change over time k, then system is called Linear Time Varying (LTV) system
- If system matrices are constant w.r.t. to time, then the system is called a Linear Time Invariant (LTI) System

$$\begin{cases} x(k+1) = Ax(k) + Bu(k), \\ y(k) = Cx(k) + Du(k) \end{cases}$$

Derivation of Solution to LTI state space system:

$$\int_{k} x(k+1) = Ax(k) + Bu(k),$$

$$y(k) = Cx(k) + Du(k)$$

• given initial state $x(0) = (\hat{x})$ and control sequence $u(0), ..., u(k), k \ge 0$, we have $x(k) = A^k \hat{x} + \sum_{j=0}^{k-1} A^{k-j-1} Bu(j)$

$$\begin{aligned} &\mathcal{K}(l) = A \times (l) + B u(b) = A \times + B u(l) \\ &\mathcal{K}(2) = A \times (l) + B u(l) = A^2 \times + A B u(b) + B u(l) \\ &\mathcal{K}(3) = A \times (2) + B u(b) = A^3 \times + A^2 B u(b) + A B u(l) + B u(l) \\ &uepends \text{ on } Ic \qquad depend \text{ on } input \text{ history.} \end{aligned}$$
For aubitrary k:

$$\begin{aligned} &\mathcal{K}(k) = A^k \times + A^{k-1} B u(b) + A^{k_2} B u(l) + \cdots + A B u(k-2) + B u(k-1) \\ &= A^k \times + \sum_{j=0}^{k-1} A^{k(j-j)} B u(j) \qquad \text{state trajectory.} \end{aligned}$$

$$\begin{aligned} &\mathcal{K}(k) = CA^k \times + \left(\sum_{j=0}^{k-1} CA^{k+1-j} B u(j) + D u(k)\right) \qquad \text{autgut trajectory.} \end{aligned}$$
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$$\begin{aligned} &2ero-input Yespense \qquad \text{CLEAR Lab @ SUSTECh} \qquad 2err-state response \end{aligned}$$

- A large portion of control applications can be transformed into a *regulation* problem
 - Regulation problem: keep certain function of the state x(k) or output y(k) close to a known constant reference value under disturbances and model uncertainties

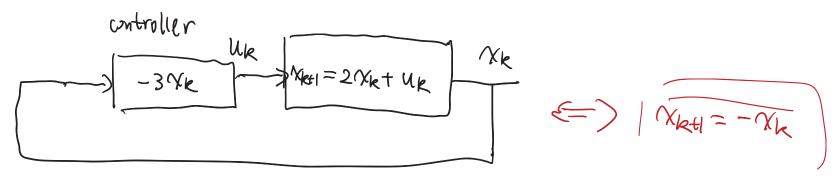




	 Keep inverted pendulum at upright position (θ = 0) Maintain a desired attitude of spacecraft or aircraft
For example:	 Air conditioner regulate temperate close to setpoint (e.g. 75F)
	 Cruise control maintain a constant speed despite uncertain road conditions
	 Converter maintains a desired voltage level for different loads

 If reference y_{ref}(t) is changing, this is no longer a regulation problem (becomes a tracking problem, which will be discussed later in this class)

- State Space Solutions • Internal Stability BIBO stability. • Controllability • Observability • Observability
- Invariance under Similarity Transformation



BIBO: bounded input bounded output stability (external stability)

• Internal Stability (with $u(k) \equiv 0$, i.e. concerned with zero-input is mainly about autonomous ystem without control or about the closed-loop system for which the controller has x(k+1) = Ax(k) + Bu(k), y(k) = Cx(k) + Du(k) alrendy been in corporated state response) Asymptotic stable: $||x(k)|| \to 0$, as $k \to \infty$, for all initial state \hat{x} • Asymptotic stable: $\|X(k)\| \rightarrow o$, fr all \hat{X} • Marginal stable: $\|X(k)\| \leq M$, for all k = 1, 2, ... $x(k) = \begin{bmatrix} \sin(\frac{2}{2}k) & \hat{X}_2 \\ 2\hat{X}_1 + (\frac{1}{2})^k \hat{X}_2 \end{bmatrix}$ • Recall state space solution for linear systems: X(k+1) = -X(k) $x(k) = A^k \hat{x} + \sum_{j=0}^{k-1} A^{k-j-1} Bu(j)$ $x(k) = a^k \hat{x} + \sum_{j=0}^{k-1} A^{k-j-1} Bu(j)$ $x(k) = a^k \hat{x} + \sum_{j=0}^{k-1} A^{k-j-1} Bu(j)$ Therefore, or linear system, the key for stability analysis is to understand how A^k behave as $k \to \infty$

$$A \in [R^{n\times n} \Rightarrow n \text{ eys} \qquad d(t(\underline{\lambda} \underline{\lambda} \underline{\lambda}) = 0)$$
• Case 1: diagonal matrix: e.g. $A = \begin{bmatrix} \lambda_1 & 0 & 0 \\ 0 & \lambda_2 & 0 \\ 0 & 0 & \lambda_3 \end{bmatrix}$

$$A^2 = \begin{bmatrix} \lambda_1^{\lambda_1^{\perp}} & 0 \\ \lambda_2^{\perp} & 0 \\ \lambda_2^{\perp} & 0 \end{bmatrix} \Rightarrow A^k = \begin{bmatrix} \lambda_1^{k} & \lambda_2^{k} \\ \lambda_2^{k} \\ \lambda_3^{k} \\ \lambda_3^{k} \end{bmatrix}$$
For internal stability, $u = 0$, $\chi(k) = A^k \cdot \chi$, so the system is asym
if $[\lambda_1| < 1, |\lambda_2| < 1, \dots, |\lambda_3| < 1, \dots, |\lambda_3| < 1, \dots, |\lambda_3| < 1, \dots, |\lambda_3| < 1 \\$
• Case 2: diagonalizable matrix, i.e. $\exists T$ such that $A = TDT^{-1}$ circle
$$D = \begin{bmatrix} \lambda_1 & \lambda_2 \\ \lambda_2 \\ \lambda_3 \end{bmatrix}$$
In this case, $A^k = TDT^{-1} TDT^{-1} \dots (TDT^{-1})$

$$= TD^kT^{-1} = TD^k \begin{bmatrix} \lambda_1^{k} \\ \lambda_2^{k} \\ \lambda_3^{k} \end{bmatrix} T^{-1}$$
• system is asym stable iff all eigs lie
inside unit circle

In all cases

<u>Theorem (Internal stability</u>): LTI (*A*, *B*) is asymptotically stable if all eigs of *A* satisfies |λ_i| < 1

ς.

- More about stability
• Asym stability
• Asym stable :
$$\chi(k) = A^{k} \chi_{0} \rightarrow 0$$
 as $k \rightarrow \infty$ for all $\chi_{0} \in |k|^{n}$
(assumed $u(k) \equiv 0$)
• stability Test: $eig(A) \in Unit Circle |eig(A)| < |$
- Most general case: A can always be written as $A = TJT^{-1}$
where J is Jordan form of A
• when A is oliagonalizable, $J = aliagonal matrix$
• ... A is not, $J = \begin{bmatrix} J_{1} \\ J_{2} \\ J_{3} \end{bmatrix} \int Jordan blocks$
• $\chi(k) = A^{k} \chi_{0} = (TJT^{-1} TJT^{-1} \dots) \chi_{0} = TJ^{k}T^{-1} \chi_{0}$

$$= T \cdot \begin{bmatrix} J_{i}^{k} & 0 \\ 0 & J_{s}^{k} \end{bmatrix} T^{-1} \cdot \chi_{0}$$

we need all Jordan blocks $\Rightarrow o$ as $k \uparrow oo$

e.g. $J = \begin{bmatrix} J_{i}^{i} & 0 & J_{s}^{i} \\ 0 & 2 & 0 & J_{s}^{i} \end{bmatrix} J_{s}$

 $i = \begin{bmatrix} 0 & 2 & 0 & J_{s}^{i} \\ 0 & 4 & 1 & 0 & 0 & 0 \\ 0 & 0 & 4 & 0 & J_{s}^{i} \end{bmatrix} J_{s}^{2} = \begin{bmatrix} \lambda_{i}^{2} & 2\lambda_{i} \\ 0 & \lambda_{i}^{2} \end{bmatrix} J_{s}^{3} = \begin{bmatrix} \lambda_{i}^{3} & 3\lambda_{i}^{2} \\ 0 & \lambda_{i}^{3} \end{bmatrix} J_{s}^{k} = \begin{bmatrix} \lambda_{i}^{k} & k & J_{s}^{k} \\ 0 & \lambda_{i}^{k} \end{bmatrix} J_{s}^{k} \Rightarrow i \text{ iff } |\lambda_{i}| < 1$

 $k \cdot \lambda_{i}^{k} = \begin{bmatrix} \lambda_{i}^{k} & k & \lambda_{i}^{k} \\ 0 & \lambda_{i}^{k} \end{bmatrix} J_{s}^{k} = \begin{bmatrix} \lambda_{i}^{k} & k & \lambda_{i}^{k} \\ 0 & \lambda_{i}^{k} \end{bmatrix} J_{s}^{k} = \begin{bmatrix} \lambda_{i}^{k} & k & \lambda_{i}^{k} \\ 0 & \lambda_{i}^{k} \end{bmatrix} J_{s}^{k} \Rightarrow i \text{ iff } |\lambda_{i}| < 1$

hilp

7_k

3x3 block:
$$J_{i} = \begin{bmatrix} \lambda_{i} & 1 & 0 \\ 0 & \lambda_{i} & 1 \\ 0 & 0 & \lambda_{i} \end{bmatrix} \xrightarrow{V_{i} \in I_{i}} \begin{bmatrix} \lambda_{i} & k & k+1 \\ \lambda_{i} & k & k+1 \\ 0 & \lambda_{i} & k \\ 0 & \lambda_{i} & k \\ 0 & 0 & \lambda_{i} \end{bmatrix}$$

$$\cdot \bigcirc is derivative of \Delta$$

$$\cdot \bigsqcup_{is} derivative of \Delta$$

$$J_{i} = 0 \quad \lambda_{i}$$

$$J_{i} = 0 \quad \lambda_{i}$$

$$J_{i} = 0 \quad \lambda_{i}$$

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- · Bounded input bounded subput (BIBO) : [bounded input Must produce bounded output]
 - · bounded signal u(h). [u(h)]=M for kEN for

Note: det(
$$zI - A$$
) = $(z - \lambda_1)(z - \lambda_2) \dots (z - \lambda_n)$
characteristic polynomial of z , det($zI - A$) = 0 has a roots that are eigs of A
 $\exists \lambda_1, \lambda_2, \dots, \lambda_n$
 $\exists All piles of H(z)$ must be eigs of $A \Rightarrow [poles(I+(z))] \geq 1$
 $(z) BIBO \Rightarrow Asym scability$
some eigs of A may not show up in the expression of $H(z)$
 $(Maybe cancelled by Co-factor of B, C)
eig.: $A = \begin{bmatrix} 0.5 & 1 \\ 0 & 3 \end{bmatrix}$, $B = \begin{bmatrix} 1 \\ 0 \end{bmatrix}$, $C = CI = 0$$

$$I_{s} :t \text{ asym } \text{ stuble } i \qquad \underline{No} \qquad e_{0}(A) = \{ o_{1}S_{1}, \{ g \} \}$$

$$- I_{s} :t \text{ BIBO?}$$

$$H(g) = C(zI - A)^{T} B \qquad (zI - A)^{T} = \left(\begin{bmatrix} 2 & 0 \\ 0 & z \end{bmatrix} - \begin{bmatrix} 0 \cdot S & 1 \\ 0 & 3 \end{bmatrix} \right)^{T} = \left(\begin{bmatrix} 2 - 0 \cdot S & -1 \\ 0 & z - 3 \end{bmatrix}^{-1} \right)^{T}$$

$$= \left(\begin{bmatrix} 1 & 2 & 0 \\ 0 & z - 3 \end{bmatrix}^{-1} \right)^{T} = \left(\begin{bmatrix} 2 - 0 \cdot S & -1 \\ 0 & z - 3 \end{bmatrix}^{-1} \right)^{T}$$

$$H(g) = (1 - 0)^{T} = \left(\begin{bmatrix} 1 & 0 \\ 0 \end{bmatrix} \right)^{T} = \left(\begin{bmatrix} 2 - 0 \cdot S & -1 \\ 0 & z - 3 \end{bmatrix}^{-1} \right)^{T}$$

$$H(g) = (1 - 0)^{T} = \left(\begin{bmatrix} 1 & 0 \\ 0 \end{bmatrix} \right)^{T} = \left(\begin{bmatrix} 2 - 0 \cdot S & -1 \\ 0 & z - 3 \end{bmatrix}^{-1} \right)^{T}$$

$$P(g) = (1 - 0)^{T} = \left(\begin{bmatrix} 1 & 0 \\ 0 \end{bmatrix} \right)^{T} = \left($$

- State Space Solutions
- Internal Stability
- Controllability
- Observability
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Nontrivial Fact (by Cayley Hamilton theorem): For a square matrix $A \in \mathbb{R}^{n \times n}$, A^k for arbitrary k can be written as a linear combination of $\{I, A, A^2, \dots, A^{n-1}\}$ $A^{k} = \alpha_{k}A^{n} + \alpha_{k-1}A^{n-1} + \dots + \alpha_{1}A + \alpha_{0}I$ = 00 ItalA+.. + dr. And • E.g.: $A = \begin{bmatrix} \cancel{1} & -1 \\ -1 & -2 \end{bmatrix}, \ \overrightarrow{A^{10}} = \begin{bmatrix} 365 & 1159 \\ 1159 & 3842 \end{bmatrix}$ n= 2 We can write $A^{10} = \alpha_0 I + \alpha_1 A = \begin{bmatrix} \alpha_0 + \alpha_1 & -\alpha_1 \\ -\alpha_1 & \alpha_0 - 2\alpha_1 \end{bmatrix} = \begin{bmatrix} 365 & 1159 \\ 1159 & 3000 \end{bmatrix}$ $\Rightarrow \alpha_1 = -1159, \quad \alpha_0 = 1524 \implies V_{i}F_{i}T_{i}$ $A = \begin{bmatrix} 123\\456\\786 \end{bmatrix}, \quad (A^{b,a}) = d \cdot I + \alpha \cdot A + d \cdot A^2$

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k-step reachability:



• Given a system (*A*, *B*), a final state x_f is called *k*-step reachable from an initial state x_0 if there exists an input sequence u(0), ..., u(k-1)such that $x(\underline{k}) = \underline{x_f}$ $@ u(\underline{k}) \in \mathbb{R}^m$

•
$$x(k) = A^{k}x_{0} + \sum_{j=0}^{k-1} A^{k-j-1}Bu(j)$$

= $A^{k}x_{0} + Bu(k-1) + ABu(k-2) + \dots + A^{k-1}Bu(0)$
• Matrix form: $X(k) - A^{k}X_{0} = \begin{bmatrix} B & AB & A^{k}B \end{bmatrix} \begin{bmatrix} u(k-1) & m-raws \\ u(k-2) & m-raws \\ u(k-2$

• Reachability Examples:
•
$$A = \begin{bmatrix} 2 & 0 \\ 0 & 2 \end{bmatrix}, B = \begin{bmatrix} 1 \\ 1 \end{bmatrix}, \text{ with } x_0 = \begin{bmatrix} 0 \\ 0 \end{bmatrix}$$
. Find out what state is reachable in 1
step, 2 steps, 10 steps?
• $A = 5 \text{tep} : x(1) = Ax_0 + But_0) = But_0$, $A_f = \begin{bmatrix} a \\ a \end{bmatrix}, from some at
2-step : $x(2) - A^2x_0 = \begin{bmatrix} B & AB \end{bmatrix} \begin{bmatrix} uu \\ u(0) \end{bmatrix} = \begin{bmatrix} 1 & 2 \\ 2 \end{bmatrix} \begin{bmatrix} uu \\ u(0) \end{bmatrix} \Rightarrow x(2) \in (cl(\begin{bmatrix} 1 & 2 \\ 1 & 2 \end{bmatrix}) = (cl(\begin{bmatrix} 1 & 2 \\ 1 & 2 \end{bmatrix}))$
 $(c-step \ reachable \ state: x_f \in (cl(\begin{bmatrix} B & AB & \cdots & A^1B \end{bmatrix}) = (cl(\begin{bmatrix} 1 \\ 1 \end{bmatrix}))$
 $(c-step \ reachable \ state: x_f \in (cl(\begin{bmatrix} B \end{bmatrix}) = IR^2, i.e. \ all \ states \ in IR^2$
 $e.g. pick, x_f = \begin{bmatrix} 1 & 2 \\ 0 & 1 \end{bmatrix}, \begin{bmatrix} 1 & 2 \\ 2 & -1 \end{bmatrix} = \begin{bmatrix} 1 & 2 \\ 0 & -1 \end{bmatrix}, (c_{LEAR \ Lob @ SUSTech}) \Rightarrow u(0) = \begin{bmatrix} 2 \\ -1 \end{bmatrix} = 14$$

Controllability

- A system matrix pair (*A*, *B*) is called **controllable** if any state $x_f \in \mathbb{R}^n$ is reachable from any initial state $x_0 \in \mathbb{R}^n$ in **finite** time steps
 - In other words, for any initial state $x_0 \in \mathbb{R}^n$ and final sate $x_f \in \mathbb{R}^n$, we can find a control input $u(\cdot)$ to steer the system from x_0 to x_f in finite time steps
- According to Reachability lemma, this requires: $x_f - A^k x_0 \in range([B, AB, ..., A^{k-1}B]), \forall x_f, x_0$ avoid the column of the colum

• So system is controllable if and only if $rank([B, AB, ..., A^{k-1}B]) = n$ for some finite k

- One way to check controllability is to keep increasing k to see whether [B, AB, ..., A^{k-1}B] is n or not
- Cayley Hamilton Theorem indicates that there is no need to check the case for k > n



 Controllability Test: (A, B) is controllable if and only if the controllability matrix $M_c \triangleq \begin{bmatrix} B & AB & \cdots & A^{\underline{n}-1}B \end{bmatrix}$, is full rank P rank $(M_c) = n$

• Cayley-Hamilton theorem implies: $rank([B \ AB \ \cdots \ A^{k-1}B]) =$ $rank([\underline{B \ AB \ \cdots \ A^{n-1}B}]), \text{ for all } k \ge n$

- controllable
$$\subseteq$$
 rank ([B AB $- A^{k-1}B]) = n$, for some finite k
Mk

Why?: suppose
$$k=n+1$$
, $M_{k}=\begin{bmatrix} B & AB & A^{n+1}B & A^{n+1}B \\ B=[b, b_{2} & b_{m}\end{bmatrix}$
We know $A^{n}=\alpha_{0}I+\alpha_{1}A+\dots+\alpha_{n-1}A^{n-1} \Rightarrow A^{n}B=[A^{n}b_{1}A^{n}b_{2}\dots A^{n}b_{m}]$

NXM

$$\Rightarrow A^n B$$
 is a linear combination of columns of M_c
 $\text{Wei Zhang} \Rightarrow (M_k) = \operatorname{rank}(M_c)$ CLEAR Lab @ SUSTech

• Examples:
•
$$A = \begin{bmatrix} 2 & 0 \\ 0 & 2 \end{bmatrix}, B = \begin{bmatrix} 1 \\ 1 \end{bmatrix}$$

• $A = \begin{bmatrix} 2 & 0 \\ 0 & 2 \end{bmatrix}, B = \begin{bmatrix} 1 \\ 1 \end{bmatrix}$
• $A = \begin{bmatrix} 1 & 2 \\ 0 & 1 \end{bmatrix}, B = \begin{bmatrix} 1 & 1 \\ 1 & 0 \end{bmatrix}$
• $A = \begin{bmatrix} 1 & 2 \\ 0 & 1 \end{bmatrix}, B = \begin{bmatrix} 1 & 1 \\ 1 & 0 \end{bmatrix}$
• $A = \begin{bmatrix} 1 & 2 \\ 0 & 1 \end{bmatrix}, B = \begin{bmatrix} 1 & 1 \\ 1 & 0 \end{bmatrix}$
• $A = \begin{bmatrix} 1 & 2 \\ 0 & 1 \end{bmatrix}, B = \begin{bmatrix} 1 & 1 \\ 1 & 0 \end{bmatrix}$
• $A = \begin{bmatrix} 1 & 2 \\ 0 & 1 \end{bmatrix}, B = \begin{bmatrix} 1 & 1 \\ 1 & 0 \end{bmatrix}$
• $A = \begin{bmatrix} 2 & 1 & 0 \\ 0 & 2 & 1 \\ 0 & 0 & 2 \end{bmatrix}, B = \begin{bmatrix} 0 \\ 1 \\ 1 \end{bmatrix}$
• $A = \begin{bmatrix} 2 & 1 & 0 \\ 0 & 2 & 1 \\ 0 & 0 & 2 \end{bmatrix}, B = \begin{bmatrix} 0 \\ 1 \\ 1 \end{bmatrix}$
• $A = \begin{bmatrix} 0 & 0 & 1 \\ 0 & 1 & 2 \\ 1 & 2 & 4 \end{bmatrix}$, $rank(M_{c}) = 3 = n \Rightarrow Controllable$
• $A = \begin{bmatrix} 0 & 0 & 1 \\ 0 & 1 & 4 \\ 1 & 2 & 4 \end{bmatrix}$, $rank(M_{c}) = 3 = n \Rightarrow Controllable$

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- Motivating aliscussim for observability
• To design state-feedback control, we need to know the 'state"
$$x$$

• but after times we can only measure autput: $y(k) = Cr(k) + Du(k)$
 1° if C is square and invertible, then $X(k) = C^{-1}(y(k) - Du(k))$
equivalent to directly measure all the state
 2° . "C" is not invertible $y(D) = C X(o) + Du(o)$
 $y(D) = C X$

Observability:

- A system (A, B, C) is called **observable** if any initial state $\underline{x}_0 \in \mathbb{R}^n$ can be uniquely determined given the system input trajectory $u(0), u(1), \dots, u(k-1)$ and output trajectory $y(0), y(1), \dots, y(k)$ for some finite k
- What is the relation between output and initial state?

• Note
$$x(k) = A^{k}x_{0} + \sum_{j=0}^{k-1} A^{k-j-1}Bu(j) \in$$

 $= A^{k}x_{0} + Bu(k-1) + ABu(k-2) + \dots + A^{k-1}Bu(0)$
 $= A^{k}x_{0} + [A^{k-1}B A^{k-2}B \dots A^{k-3}B 3] (u(0) = m \times 1 \quad \text{Vector}$
 $M \cdot (columns)$
 $y(0) = (X(0) + Du(0) = (A_{X(0)} + C Bu(0) + Du(1))$
 $y(1) = (X(1) + Du(1) = (A_{X(0)} + C Bu(0) + Du(1))$
 $y(2) = (Y(2) + Du(2) = (A^{2}_{X(0)}) + C AB u(0) + C Bu(1) + Du(0)$
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TR Continue derivation for relation between $y(\cdot)$ and x_0 $T_{k} = \begin{bmatrix} y(e) \\ y(e)$ U() Let's define Uk= 4(1) 1 Ly(m) 4(4-1) Ulo) unknown D CB DOD... Y() yu) k(1)CA 4(2) Xo/ CA2 y(2) . L y(k1) u(k) CĄ (1.7)×n TRE known UREL CLEAR Lab @ SUSTee Wei Zhang

T K. unknow bnaun • In vector form, we have: $Y_k = O_k x_0 + T_k U_k \Rightarrow O_k x_0 = (Y_k - T_k U_k)$ • O_k maps initial state x_0 to ouput over time [0, k - 1]• T_k maps input to output over time [0, k-1] for example : $Q_k \times = b$ VX ENuil (Or) • $Null(O_k)$ gives ambiguity in determining x_0 $Q_{k} \stackrel{\sim}{x} = 0 \Rightarrow O_{k}(x, +\tilde{x}) = b$ x_0 can be uniquely determined if $Null(O_k) = \{0\}$, i.e. $rank(O_k) = n$, for some finite *k* Recall : conservation of dim $dim(Null(Q_k)) + rank(Q_k) = n$

- Input *u* does not affect **ability** to determine x₀
 - its effect can be subtracted out
 - So we often say system (*A*, *C*) is observable or not (No need to mention *B*)

$$\begin{aligned} & \left\{k = \begin{bmatrix} c_{A} \\ c_{A} kr \end{bmatrix}, k \right\} \\ \bullet & Observability Test: (A,C) is observable if and only if observability matrix M_{0} is full rank, where $M_{0} = \begin{bmatrix} c_{A} \\ c_{A} \\ c_{A} \end{bmatrix} \\ \bullet & Note: (A,C) observable \Leftrightarrow Any x_{0} can be uniquely determined by $u(0), ..., u(k-1), y(0), ..., y(k-1), \text{ for a finite } k \\ u(0), ..., u(k-1), y(0), ..., y(k-1), \text{ for a finite } k \\ u(0), ..., u(k-1), y(0), ..., y(k-1), \text{ for a finite } k \\ u(0), ..., u(k-1), y(0), ..., y(k-1), \text{ for a finite } k \\ u(0), ..., u(k-1), y(0), ..., y(k-1), \text{ for a finite } k \\ u(0), ..., u(k-1), y(0), ..., y(k-1), \text{ for a finite } k \\ u(0), ..., u(k-1), y(0), ..., y(k-1), \text{ for a finite } k \\ u(0), ..., u(k-1), y(0), ..., y(k-1), \text{ for a finite } k \\ u(0), ..., u(k-1), y(0), ..., y(k-1), \text{ for a finite } k \\ u(0), ..., u(k-1), y(0), ..., y(k-1), \text{ for a finite } k \\ u(0), ..., u(k-1), y(0), ..., y(k-1), \text{ for a finite } k \\ u(0), ..., u(k-1), y(0), ..., y(k-1), \text{ for some finite } k \\ u(0), ..., u(k-1), y(0), ..., y(k-1), u(k-1), u(k$$$$

- For case
$$(1): M_0 = \begin{bmatrix} C \\ CA \end{bmatrix} = \begin{bmatrix} 1 & 0 \\ 1 & ot \end{bmatrix}$$
 Vauk $(M_0) = 2$
 $h=2$ $(A) = \begin{bmatrix} 1 & 0 \\ 1 & ot \end{bmatrix}$ \Rightarrow observable

For case
$$(2)$$
: $M_{0} = \begin{bmatrix} C \\ CA \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ 0 & 1 \end{bmatrix}$ vou $k(M_{0}) = \begin{bmatrix} C \\ CA \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ 0 & 1 \end{bmatrix}$ in observable

try to have more rows

$$O_{K} = \begin{bmatrix} C \\ CA \\ CA^{2} \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ 0 & 1 \\ 0 & 1 \end{bmatrix} + does of t have to have$$

more measurement

- State Space Solutions
- Internal Stability /
 Controllability /
 Observability /

- Invariance under Similarity Transformation

Change of basis vectors for equivalent system representation

• Canonical coordinate system:
$$\chi = \begin{pmatrix} 2 \\ 3 \end{pmatrix} = 2 \cdot e_1 + 3 \cdot e_2$$

New coordinate system: basis vectors

$$P_{1} = \begin{bmatrix} 1 \\ 1 \end{bmatrix}, P_{2} = \begin{bmatrix} -1 \\ 2 \end{bmatrix}, P_{2}$$

- Dynamics in new coordinate systems:
 - $x(k) = P\underline{z}(k) \Rightarrow z(k) = P^{-1}x(k)$

Dynamics in original coordinate:

$$x(k+1) = Ax(k) + Bu(k), \quad y(k) = Cx(k) + Du(k),$$

Dynamics in the new coordinate

y(k) = (c) p(k) + (D) u(k)

$$\int z(k+1) = \hat{A}z(k) + \hat{B}u(k), \ y(k) = \hat{C}z(k) + \hat{D}u(k)$$

Derive $(\hat{A}, \hat{B}, \hat{C}, \hat{D})$

$$z(k+1) = p^{-1} \underline{x}(k+1) = p^{-1} A \underline{x}(k) + p^{-1} B u(k)$$

$$= p^{-1} A p^{-1} \underline{x}(k) + p^{-1} B u(k)$$

$$\hat{B} = p^{-1} B$$

Â

• System $(\hat{A}, \hat{B}, \hat{C}, \hat{D})$ is called **equivalent** to original system (A, B, C, D) f defined in O for any nonsingular p 26

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B

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- Similarity transformation does not change system stability, controllability and observability
 - A and PAP⁻¹ have the same Jordan form, so similarity transformation does not change stability, i.e., <u>A is (asymptotically)</u>
 stable iff is (asymptotically) stable

General case:
$$A = TJT^{-1}$$
, Fridan form
 $\widehat{A} = PAP^{-1} = PTJT^{-1}P^{-1} = (PT)J.(pT)^{-1}$
 $\widehat{A} \sim A$ has the same set of eigenvalues

• $\hat{M}_c = P^{-1}M_c$, with this it can be shown that M_c full rank iff \hat{M}_c full rank

therefore, (A, B) controllable iff $(\widehat{A}, \widehat{B})$ controllable

$$\widehat{M}_{c} = [\widehat{B} \quad \widehat{A} \widehat{B} \quad \cdots \quad \widehat{A}^{n_{1}} \widehat{B}]$$

$$= [P^{T}B \quad P^{t}APP^{T}B \quad \cdots \quad (P^{t}AP)^{n_{1}} P^{t}B]$$

$$= P^{T} [B \quad AB \quad \cdots \quad A^{n_{1}}B] = P^{T} M_{c}$$

$$V.F.Y. : \quad p \quad if \quad P \quad is \quad nonsingular, \quad rank(M_{c}) = rank(P^{t}M_{c}) \leftarrow$$

$$\widehat{M}_{o} = M_{o}P, \text{ similarly, this implies } (A, C) \text{ observable iff } (\widehat{A}, \widehat{C}) \text{ observable}$$

$$rauk(\widehat{M}_{o}) = n \quad (=) \quad runk(M_{o}) = n$$

$$M_{o} \cdot P$$

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